Investigations on current mills and wind turbines



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Investigations on current mills and wind turbines

- Introduction
- Boundary Conditions
 - wind vs. tidal currents
- Theory of energy conversion
 - Betz' theory
 - airfoil theory lift and drag
- vertical axis free stream turbines
 - vertical vs. horizontal axis
 - drag vs. lift based designs
 - analytical modell for lift based design problems
 - pitch-variable design use of drag
 - computational methods for drag based design at low TSR







•The energy situation is getting more difficult





• the will to reduce greengas emissions



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possible alternatives:

- e.g. nuclear power
- renewable energy resources







wind generation:

- due to geographical diffenrences, there are developing areas of our atmosphere of higher and some of lower pressure







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development of tidal currents: - cyclic changes in the sea level - due to gravitation forces in the interplanetary system Sun – Moon - Earth





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examples:











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wind vs. tidal currents:

	wind	tidal currents
density p	1,2 kg/m³	1000-1020 kg/m³
viskosity η	0,0000171 N/(m² s)	0,001 N/(m² s)
velocity v	10 m/s	2 – 3 m/s
Re	300000	450000

$$Re=\rho v L \eta^{-1}$$





wind vs tidal currents

- same Re – numbers in wind and water:



we can use technologies proved in wind energy to convert energy of tidal

streams

- water and wind streams through free stream turbines can both be calculated incompressible



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major advantage of tidal currents in comparison to wind streams:

- cyclic variation of sealevel and the velocity of the stream are **predictable**





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conclusions:

direction of the stream changes twice a day
 machine should not be sensitive to see level change / vertical velocity change

possible solution:

- -vertical axis free stream turbine
- converts power from every stream direction
 - this compensates the lower efficiency
- generator and transmission could be built over the water surface



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Betz' theory:

total wind power through a control area A :

$$P_{wind} = \frac{dE_{wind}}{dt} = \frac{d}{dt}(\frac{1}{2}mv_{\infty}^2) = \frac{1}{2}\rho A v_{\infty}^3$$





Equation of Continuity:

 $\rho v_1 A_1 = \rho v_2 A_2 = \rho v_3 A_3$

Energy balance:

$$\Delta E = E_1 - E_3 = \frac{1}{2}m(v_1^2 - V_3^2)$$



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Betz' theory:

Power extracted in the Turbine area:

$$P_{turbine} = \dot{E} = \frac{1}{2}\dot{m}(v_1^2 - V_3^2)$$

mass flow rate through the turbine:

$$\dot{m} = \rho v_2 A_2$$

with
$$v_2 = \frac{v_1 + v_3}{2}$$
 Froude- Rankine-theorem

Power extracted by the turbine:

$$\dot{E} = \frac{1}{2}\rho \frac{v_1 + v_3}{2} A_2(v_1 + v_3)(v_1 - v_3)$$

in relation to the windenergy:

$$\dot{E} = \underbrace{\frac{1}{2}\rho A_2 v_1^3}_{P_{wind}} \underbrace{\frac{1}{2} \left(1 + \frac{v_3}{v_1} - \frac{v_3^2}{v_1^2} - \frac{v_3^3}{v_1^3} \right)}_{c_P}$$



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Betz' theory:





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vertical vs. horizontal axis:

horizontal axis machines:

-significant losses due to angular momentum at low velocities





vertical vs. horizontal axis:





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drag based turbines:

-oldest and simplest way to extract energy from the wind



anemometer



persian wind turbine



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drag based turbines:

theory: - area A with moving with velocity u

- wind speed v

Power:
$$P_T = F_W u$$
 Force: $F_W = \frac{\rho}{2} c_W A v_{rel}^2$





$$P_T = \frac{\rho}{2} c_W A (v_\infty - u)^2 u$$



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drag based turbines:





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Lift based turbines:



Darrieus turbine

H-Darrieus turbine



How can we calculate the power of such a machine?



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lift based design:





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lift based design: theory:

velocity:

 $\tan\gamma=\frac{u}{v_{\infty}}=\lambda$

power:

$$P = (F_A \cos \gamma - F_W \sin \gamma)u$$



$$P = rac{
ho}{2} A v_{\infty}^3 c_A \lambda \sqrt{1 + \lambda^2} \left(1 - rac{\lambda c_W}{c_A}
ight)$$

power coefficient:

$$c_{P,A} = c_A \lambda \sqrt{1 + \lambda^2} \left(1 - rac{\lambda}{rac{c_A}{c_W}}
ight)$$



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lift based design: theory:





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lift based design: comparison to drag based design:

- ratio of the maximum power:

$$rac{P_{max,A}}{P_{max,W}}\sim (rac{c_A}{c_W})^2$$

example: plate with glide ratio 10

use of lift is 100 times better than use of drag!





Lift based turbines:

main characteristics:

tip speed ratio:

$$\lambda = \frac{\Omega \ R}{v_1}$$

angle of attack:

$$(\Theta, \lambda) = \begin{cases} \arcsin\left(\frac{\cos\Theta}{\lambda}\right) & , \frac{\pi}{2} < \Theta < \frac{3\pi}{2} \\ -\arcsin\left(\frac{\cos\Theta}{\lambda}\right) & , sonst \end{cases}$$

relative velocity:

$$(\Theta, \lambda) = \sqrt{(v_2 + \lambda \cdot v_2 \cdot \sin\Theta)^2 + (\lambda \cdot v_2 \cdot \cos\Theta)^2}$$



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Lift based turbines:





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Lift based turbines:

Lift and drag coefficients:



approximation:

$$c_a(\Theta, \lambda) = 2 \cdot \pi \cdot \sin(\alpha(\Theta, \lambda))$$



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$$c_w(\Theta, \lambda) = \frac{c_a(\Theta, \lambda)}{100} + 0.04$$

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Lift based turbines:

Our goal: - power calculation of the turbine:





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arms of the lever $r_a = R \sin(\alpha(\Theta, \lambda))$ $r_w = R \cos(\alpha(\Theta, \lambda))$

lift based turbine: obtain the power

 $P_{Rotor}(\lambda) = n \ M_{Rotor,ges}(\lambda) \ \Omega$





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lift based design:

assumptions made:

- -homogeneous stream through the turbine
- same velocity at all blades in every position

-angular momentum is to be neglected

validation of the method:

- -influences of airfoil wake and angular momentum are essiential
- no possible at low tip speed ratio!



this modell just for general lay out (order of magnitude)





lift based design: <u>behaviour at low TSR:</u>





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lift based design: behaviour at low TSR:





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lift based design: <u>behaviour at low TSR:</u>

real power at low TSR:



-negative power at low velocities due to stall effects - no lift but still drag

- turbine will not be able to start by itself

- but: drag based systems have good starting

Idea: use pitch control to use airfoils as drag areas!



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pitch-variable design – use of drag:

- two possible kinematics:



FLM

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numeric simulation for drag based design at low TSR:





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numeric simulation for drag based design at low TSR:

results: - videos



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Thank you very much!



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Appendix: Incompressibility

pressure diffence in front of the turbine and after it is negligible

∆p≈ 0



 $\Delta \rho \approx 0$ incompressibility of the stream through the turbine

 $\delta \rho / \delta t = \delta \rho / \delta x_i = 0$



FLM

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Appendix: Froude-Rankine-Theorem

Froude-Rankine-theorem :





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Appendix: Schmitz' theory

Schmitz' theory of losses due to the angular momentum:

$$c_{P_{Schmitz}} = \int_{0}^{1} 4\lambda \left(\frac{r}{R}\right)^{2} \frac{\sin^{3}\left(\frac{2\arctan\left(\frac{R}{\lambda r}\right)}{3}\right)}{\sin^{2}\left(\arctan\left(\frac{R}{\lambda r}\right)\right)} d\left(\frac{r}{R}\right)$$



